

"A systematic approach for aircraft on-board system architecture within DEFAINE"

Christopher Jouannet Saab Aeronautics

Ingo Staack LiU

Max Baan ParaPy



17th MODPROD Workshop 7-8th Febr 2023, Linköping, Sweden

Agenda

- DEFAINE overview
- On-Board System use case
 - Architecture generation
 - Time domain model automation
- Conclusion and future work



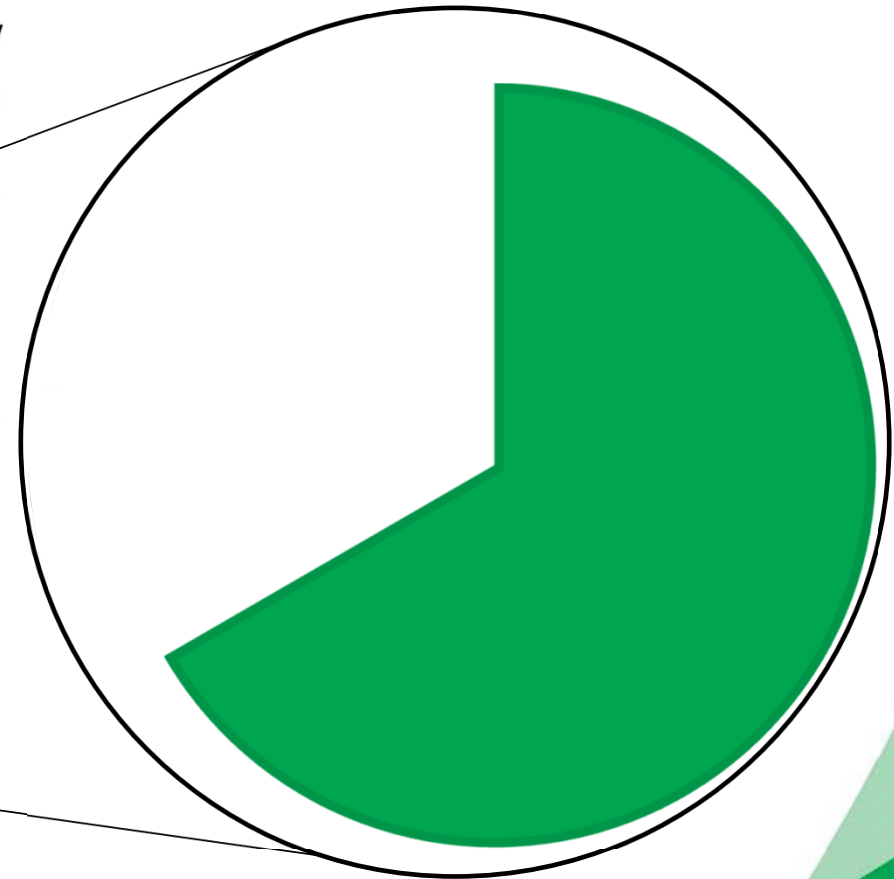
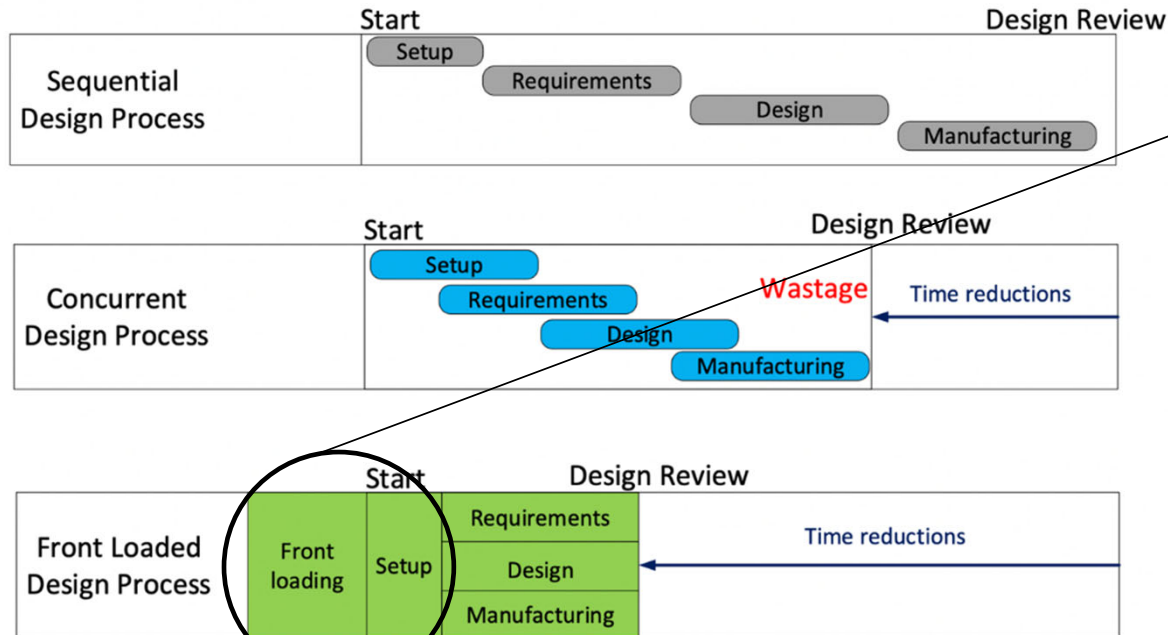
DEFAINE Objectives

- To reduce recurring cost in the design of aerospace systems and reduce the lead-time for design updates
- By enabling AI-powered front-loading
- Via a software framework that allows design engineers to perform large-scale design exploration studies

- Goals:
 - 10% cost reduction during design phase of aerospace systems
 - 50% lead-time reduction for design updates



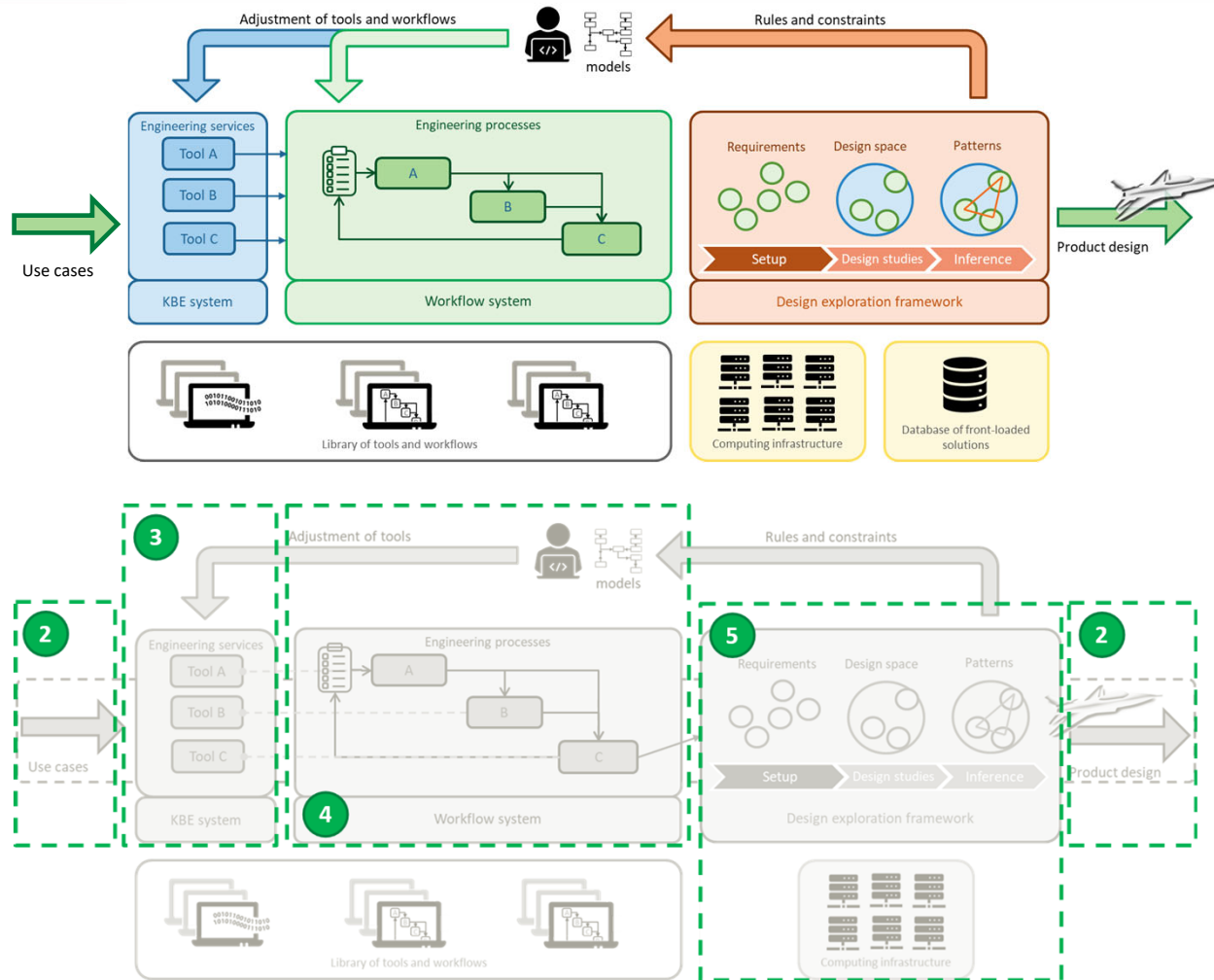
DEFAINE goals: Front loading



■ App development



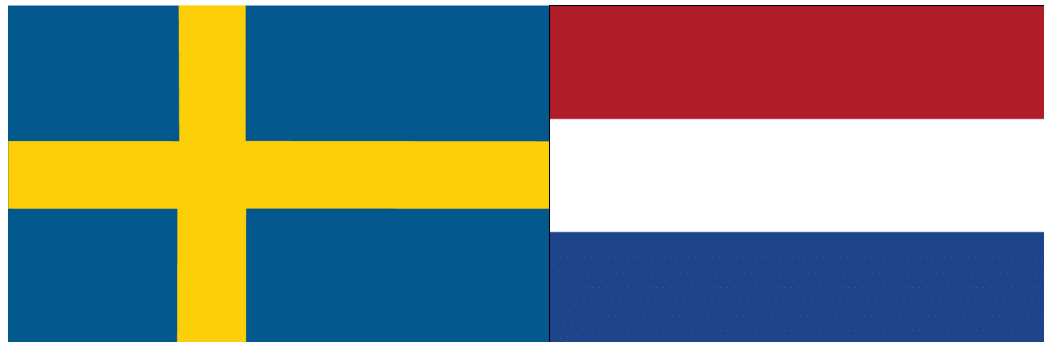
DEFAINE Overview



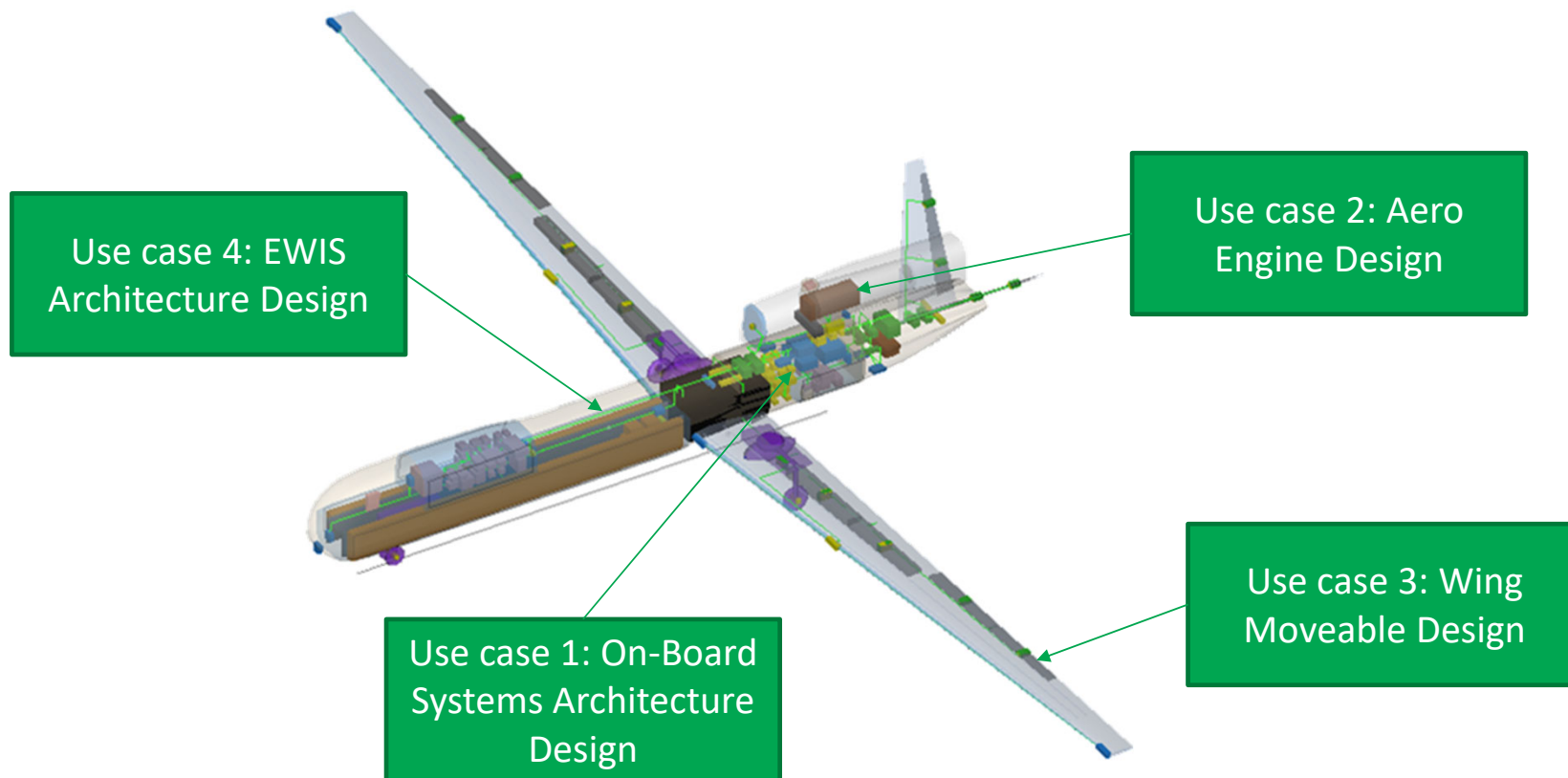
- WP 2. Industrial use cases
- WP 3. Engineering services and development methodologies
- WP 4. Workflows and data exchange standards
- WP 5. Design space exploration and AI
- WP 1: Project Coordination
- WP 6: Dissemination and exploitation



DEFAINE Partners



Industrial use cases



On-Board system use case



17th MODPROD Workshop 7-8th Febr 2023, Linköping, Sweden

Background & Use-Case Description

Background:

- UAV overall design
→ early design stage
- Current development process based on point assessment / steady-state models
- Engineering tool/environment: PaceLab®
- Main Objective:
 - DSE for global vehicle configuration
 - Selection/Combination of on-board systems

Use Case Objective:

- Enable higher fidelity dynamic model of the on-board system, based on the design/vehicle information defined within the PaceLab®-based ACD process.
- By making use of a semi-automated KBE process containing:
 - a) system architecture design inf.,
 - b) subsystem sizing (domain specific), and
 - c) simulation tool specific information



Background: Aircraft On-board Systems

- Power systems
- Specific domains: hydraulic, pneumatic, electric, (mechanic)
- System interdependencies
- Redundancy / Reliability
- Systematic split-up:
 - power generation (PGS)
 - power distribution (PDS)
 - power consumption (PCS)

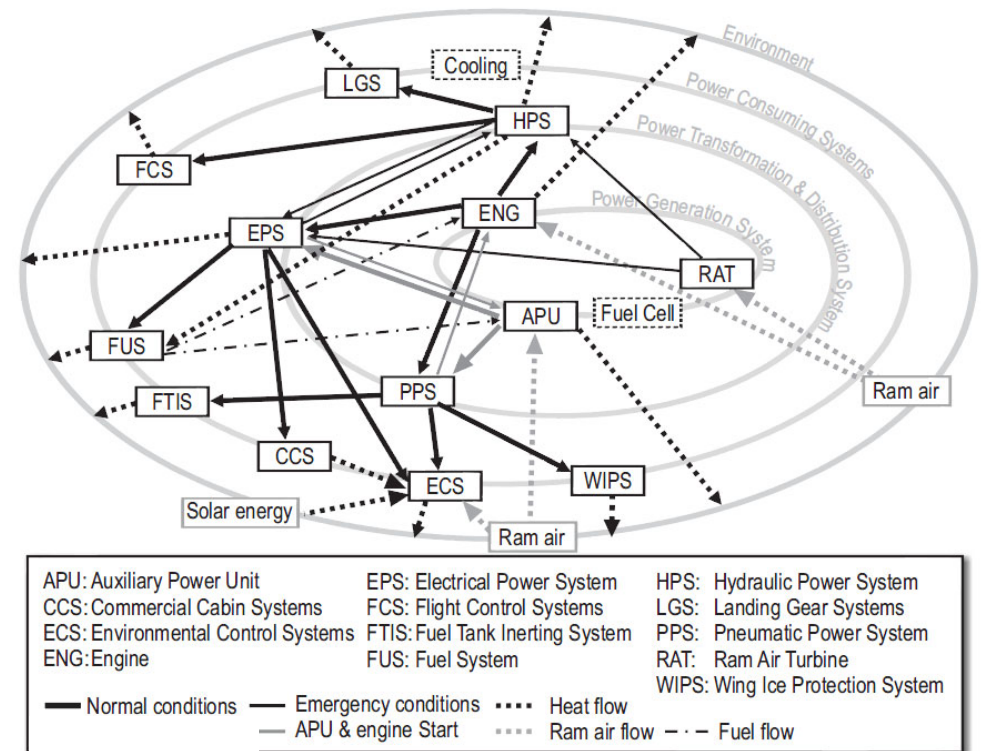


Figure 3.3: Energy flow overview for a conventional power systems architecture

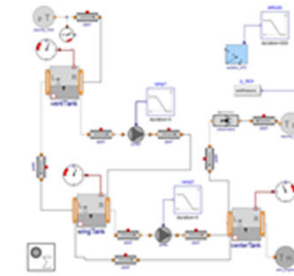
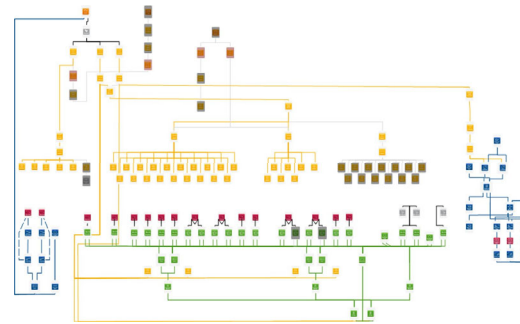
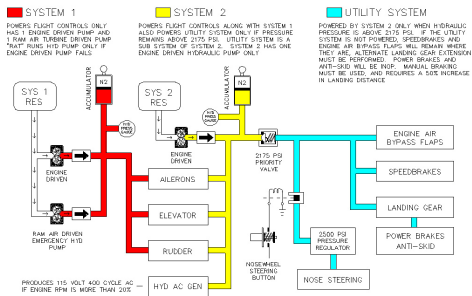
source: Liscouët-Hanke, *A model-based methodology for integrated preliminary sizing and analysis of aircraft power system architecture*, 2008.

On-Board Systems Architecture Design

- Creation of architecture is done manually
 - Time consuming
 - None added value
 - Prone to error
- Implementation of the different architectures is done manually component per component...
- Automate architecture creation and instantiation
 - KBE rules
 - Ease of implementation
 - Ease of modification



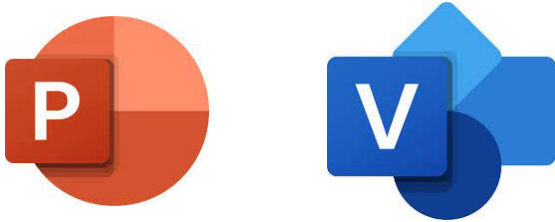
Current way of working



Architecture and technology

Steady State

Time Domain

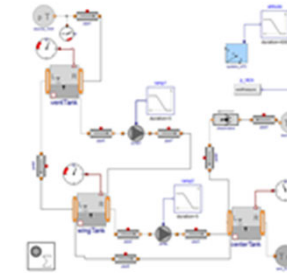
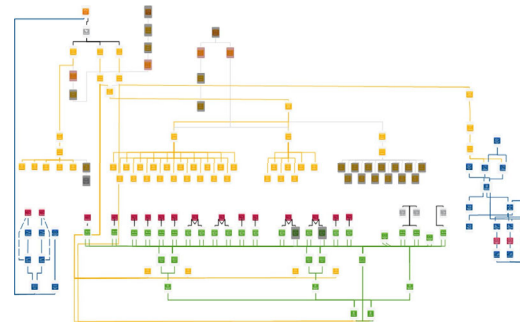
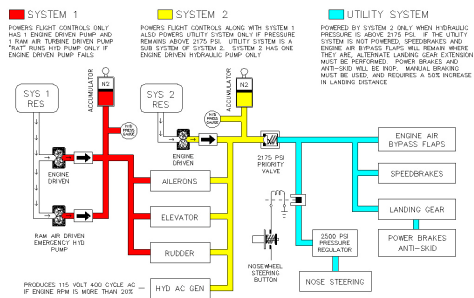


On-Board system architecture automation



17th MODPROD Workshop 7-8th Febr 2023, Linköping, Sweden

Envisioned way of working



Architecture and technology

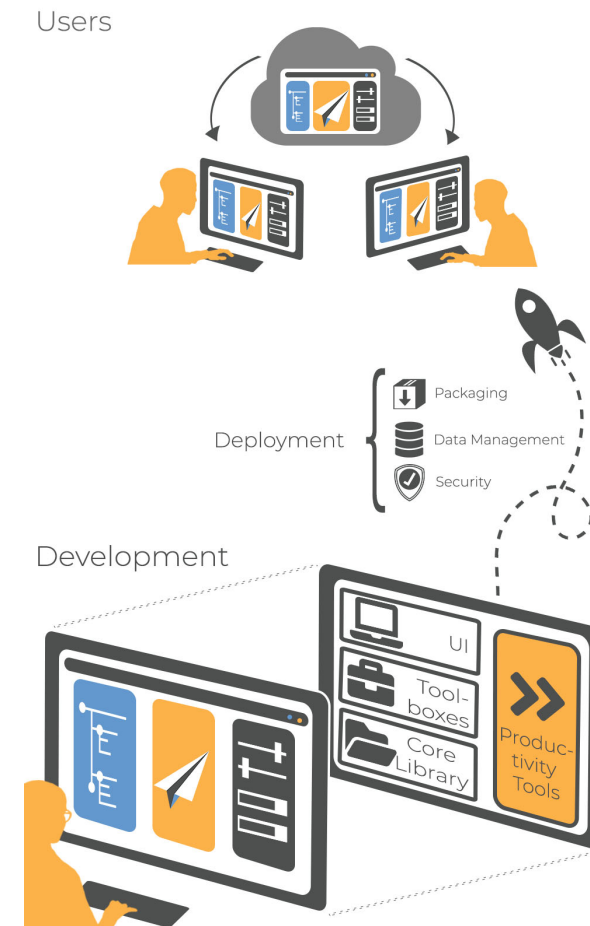
Steady State

Time Domain

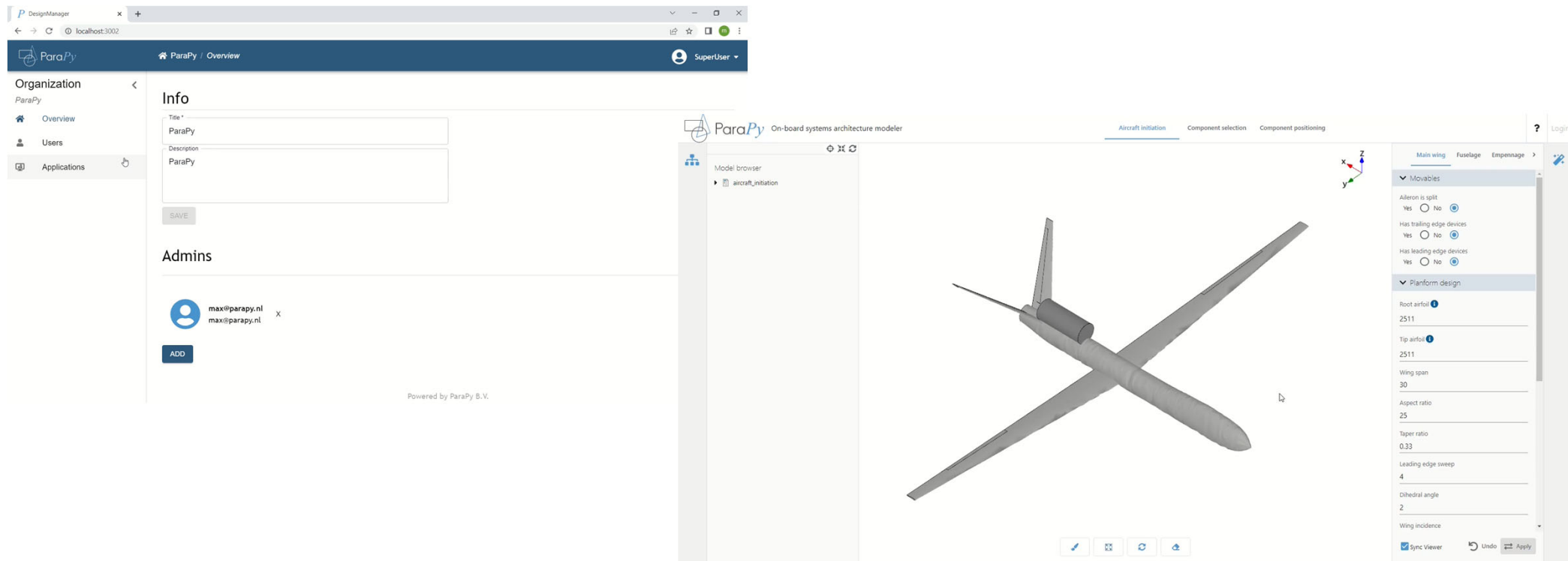


Approach

- Define logic for each type of system
 - Rules
 - Logical step
 - Input format to PaceSysArc
- Make usage of KBE environment offer by ParaPy
 - Ease of KBE app development
 - Ease of deployment: cloud



Assystor app overview



The screenshot displays the Assystor application interface, which is divided into two main sections: organization management and aircraft modeling.

Organization Management (Left Panel):

- Organization:** Shows the organization name "ParaPy" and a "SAVE" button.
- Admins:** Lists the administrator "max@parapy.nl" with an "ADD" button.
- Footer:** "Powered by ParaPy B.V."

Aircraft Modeling (Right Panel):

- Model browser:** Shows a tree view with "aircraft_initiation".
- 3D View:** Displays a 3D model of an aircraft in a perspective view.
- Properties Panel (Right):**
 - Movables:**
 - Aileron is split: Yes No
 - Has trailing edge devices: Yes No
 - Has leading edge devices: Yes No
 - Planform design:**
 - Root airfoil: 2511
 - Tip airfoil: 2511
 - Wing span: 30
 - Aspect ratio: 25
 - Taper ratio: 0.33
 - Leading edge sweep: 4
 - Dihedral angle: 2
 - Wing incidence: Sync Viewer



Current achievement architecture generation

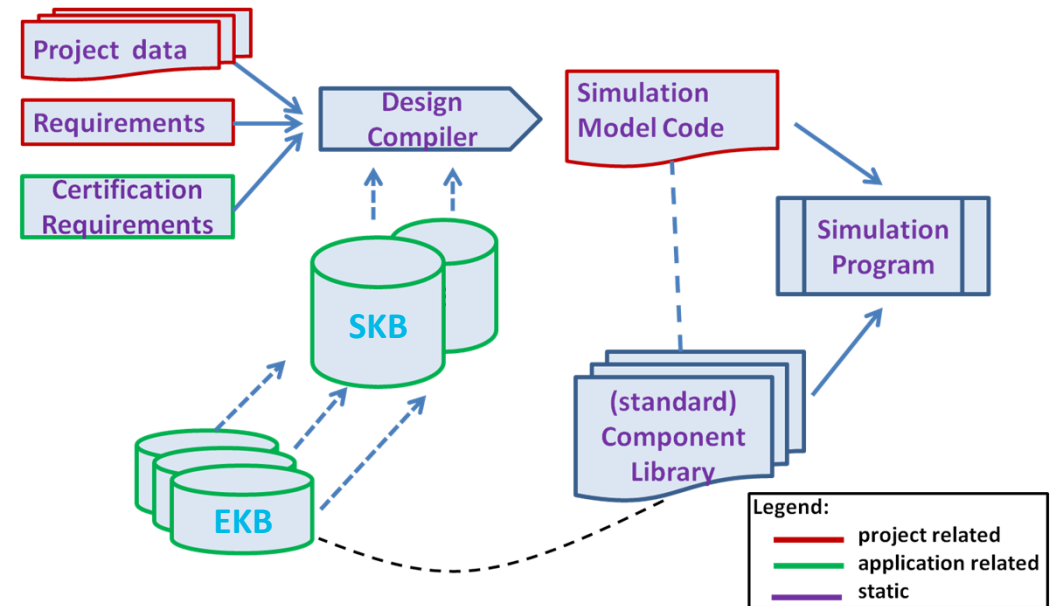
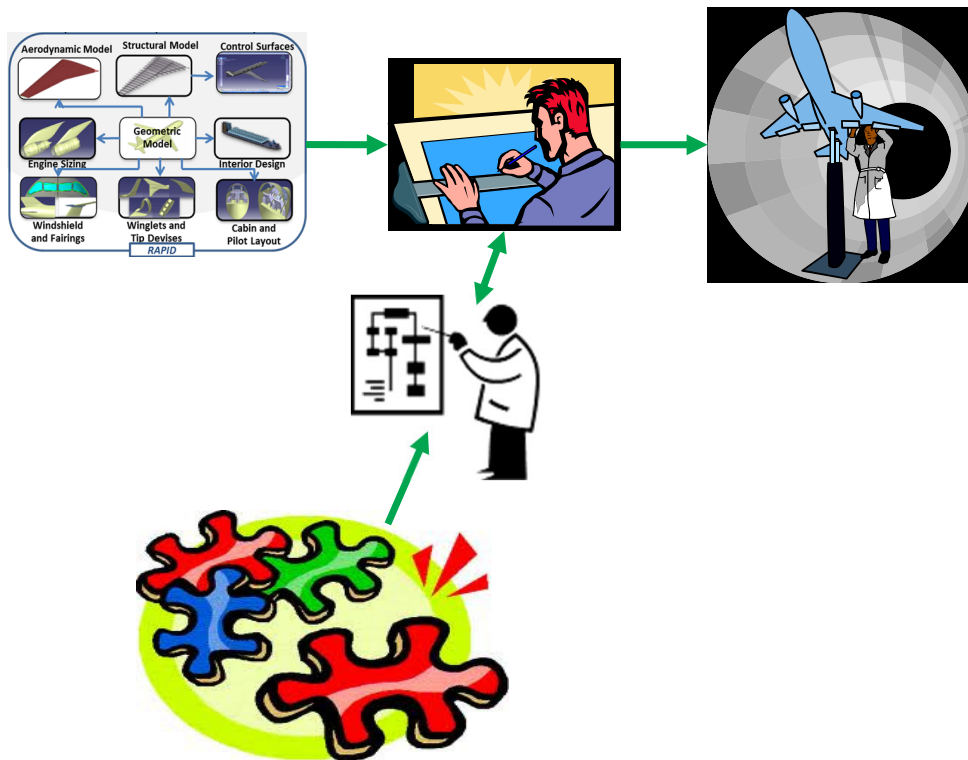
- Automatic generation of FCS architecture with regards to:
 - Nb of control surfaces
 - Redundance level
 - Technology choices
- Time reduction
 - About 40 times quicker so far:
 - Still some manual work at the import in Pace SysArc
 - Enables quicker exploration of different architectures
- No full automation of architecture generation to design space exploration realised yet
 - Will lead to further reduce of time
 - Automated import will further reduce the time from architecture creation to usage with a factor of ~30



III. KBE-based process to enable higher-fidelity (dynamic) model in the early design stage



General Approach: Tool-specific Simulation-integrating KBE Approach



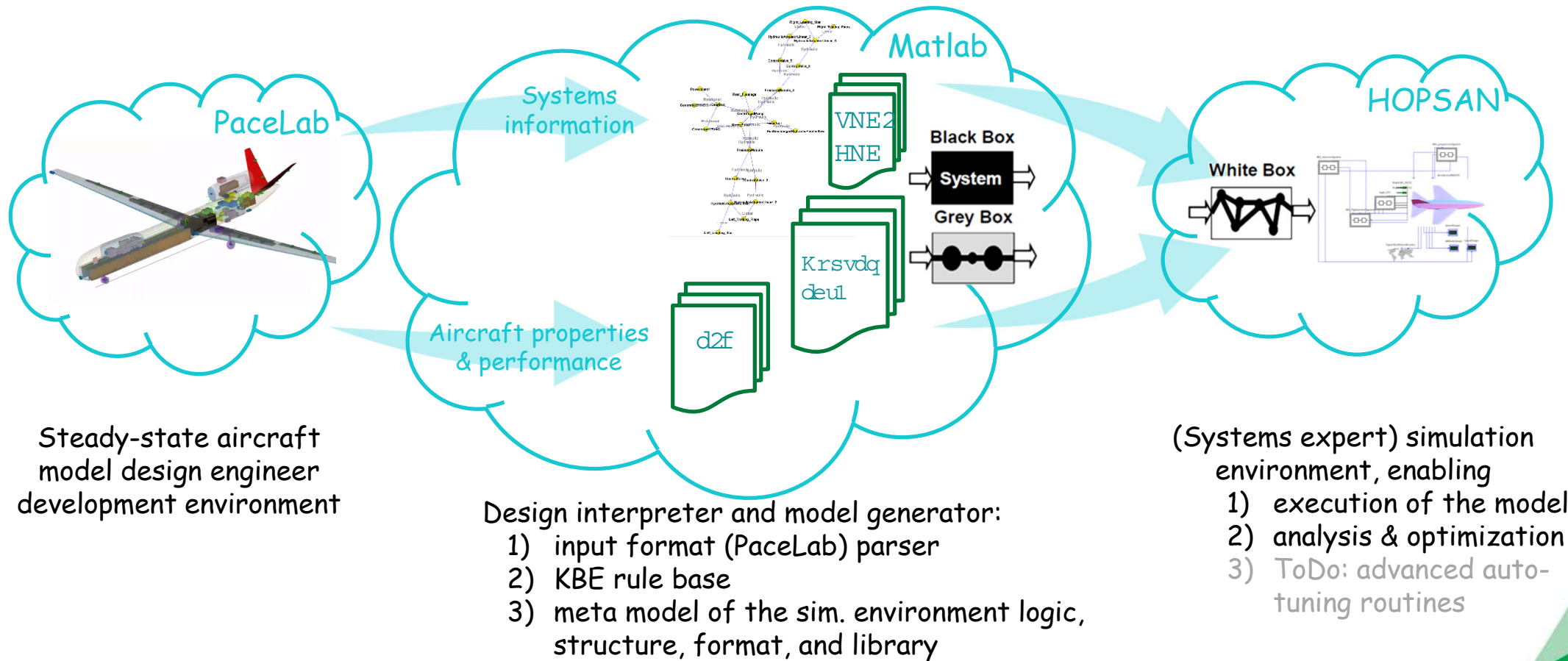
SKB: System Knowledge Base

EKB: Element Knowledge Base

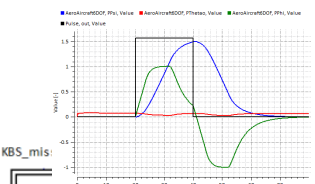
- serve for the translation from meta-components towards the simulation components in the library



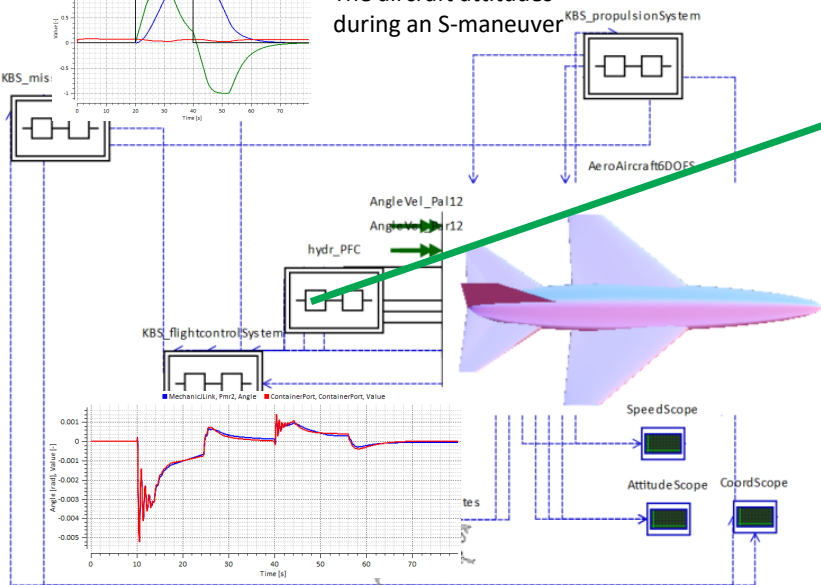
Use-Case Specific Implementation



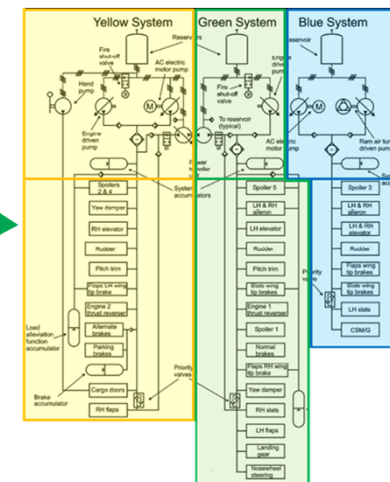
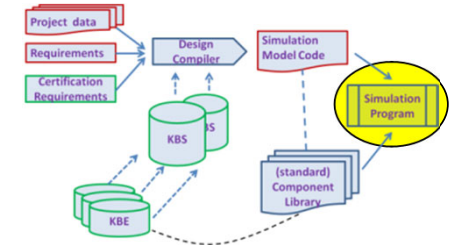
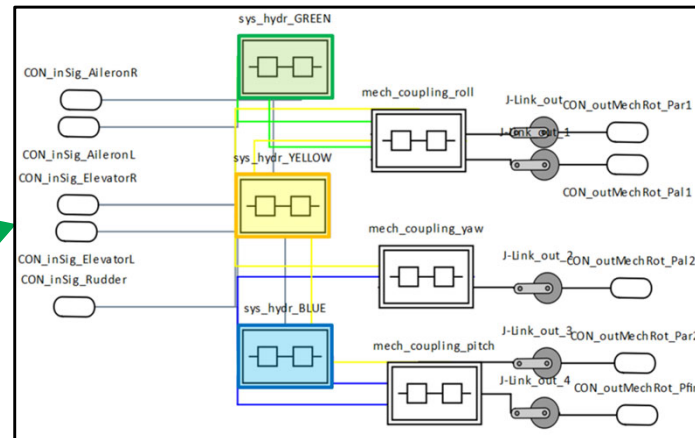
Realized Simulation Model



The aircraft attitudes during an S-maneuver



Angular position and reference position of the rudder actuator



Conclusion

- A prototype to automate the on-board system architecture creation has been created and validated
 - Showing a time reduction of more than 40 times!
 - Further development is needed to cover more systems
- Integration to the time domain is under development
 - Extraction of flight condition specific information
 - Transformation of the steady state model to time domain
 - Enabling interoperability through standard (SSP, FMU/FMI)

